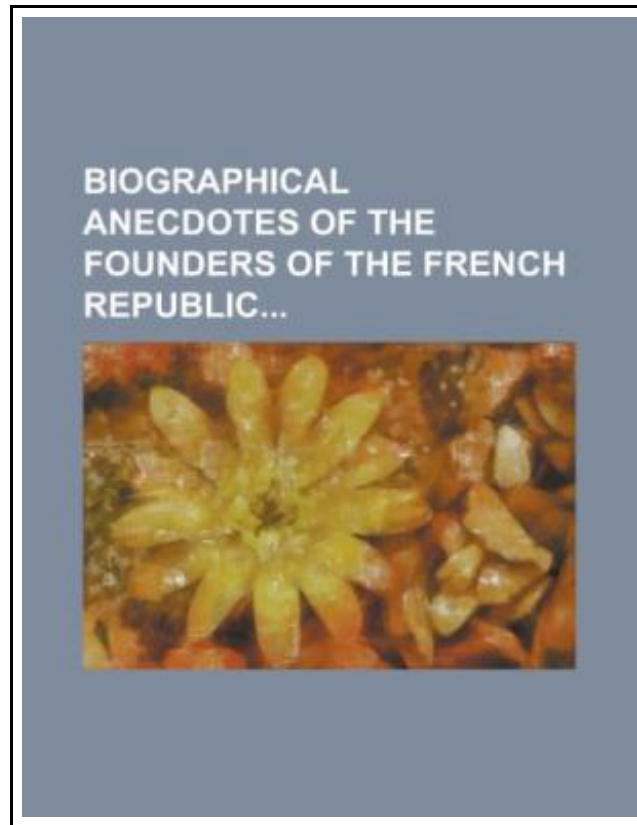


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RareBooksClub. Paperback. Book Condition: New. This item is printed on demand. Paperback. 24 pages. Original publisher: Cleveland, Ohio : National Aeronautics and Space Administration, Glenn Research Center ; Hanover, MD : Available from NASA Center for Aerospace Information ; Springfield, VA : National Technical Information Service distributor, 2003 OCLC Number: (OCoLC)56822744 Excerpt: . . . The burner section consists of an air plenum, a centrally located injection nozzle and an axial 2 is swirler, which sits coaxially around the injector. overlap provided between the A combustor and the burner section that the complete flame is captured by the viewing so of windows. The plenum has been fabricated to accommodate various sizes swirlers and injector nozzles. It is also equipped with two microphone taps for measuring forced velocity perturbations, and two water taps to provide cooling to the face plate on which the combustor sits. The face plate has a diverging quarl and is replaceable. Both pressure atomizing and air is assist atomizers can be used. The fuel (and air in case air assist nozzle) fed to the injector of nozzle from the bottom of the rig through stainless steel tubing. Axial location of the injector YI of nozzle can be changed by adjusting the position the lance, which houses the nozzle. A variety axial swirlers can be used with different vane angles and sizes. Their location can also be of changed along the longitudinal axis of the burner. Slgnak Tanperrtura Computer Dlgltal Camera Cornburtor Alr Campmrtror A Flow Control. Q. Computmr Cilmp Matar Tank Fuel 9: of Figure Schematic Overall Combustor Test Facility 9. The overall combustion facility layout is shown in Figure Kerosene is used as the combustion 885X fuel and is delivered from the non-pressurized tank to the combustor by a SUNTEC Model 300 rpm of...



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